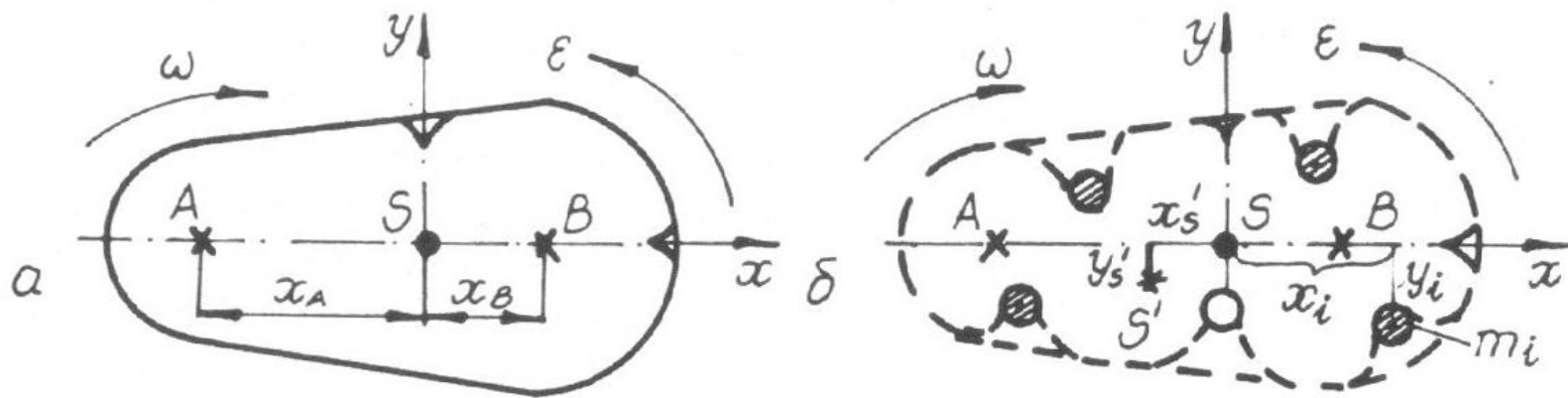


Уравновесяване на масови СИЛИ

Точкови модели

Точкови модели на звена и механизми. В теоретичната механика се доказва, че при общо равнинно движение на всяка звено (тяло) разпределените по цялото равнинно сечение на звеното елементарни инерционни сили се редуцират спрямо масовия център до една динама \vec{F} , \vec{M}_F . И също, че всяка равнинна система сили се редуцира в общия случай до една динама. Оттук следва, че равнинната система от безкрайно много елементарни инерционни сили на звеното може да се замени с други системи сили, еквивалентни на нея по отношение на резултантната динама. За практически цели е удобно еквивалентната система да има ограничен брой крайни по големина инерционни сили.



$$\vec{\Phi} = -m\vec{a}_s$$

$$\vec{M}_\Phi = -J_s\vec{\epsilon}$$

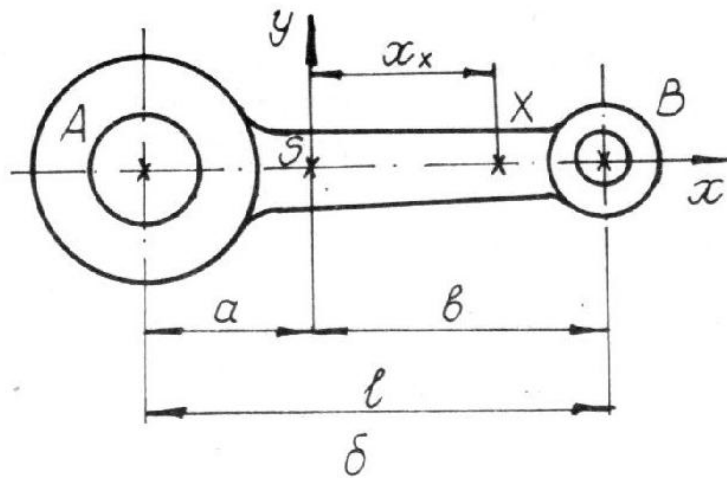
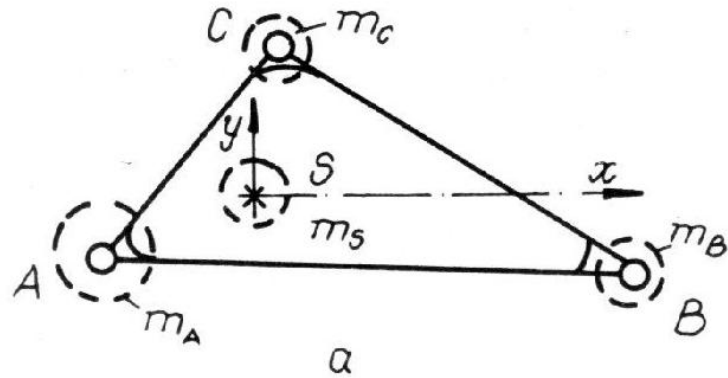
$$\vec{\Phi}' = -m' \vec{a}'_s; \quad m' x'_s = \sum_{i=1}^k m_i x_i; \quad m' y'_s = \sum_{i=1}^k m_i y_i,$$

$$\sum_{i=1}^k m_i = m; \quad \sum_{i=1}^k m_i x_i = 0; \quad \sum_{i=1}^k m_i y_i = 0.$$

$$M'_\phi = -J'_s \varepsilon = -\varepsilon \sum_{i=1}^k m_i (x_i^2 + y_i^2).$$

$$\sum_{i=1}^k m_i (x_i^2 + y_i^2) = J_s$$

$$E = m \frac{V_s^2}{2} + J_s \frac{\omega^2}{2}.$$

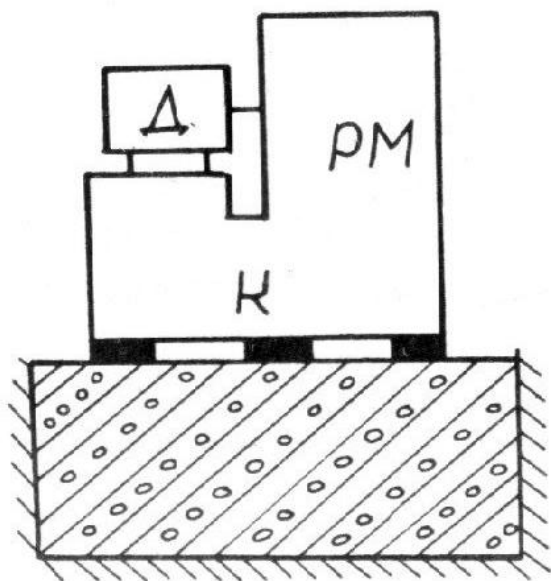


$$\begin{cases} m_A + m_s + m_B = m \\ -m_A a + m_B b = 0 \\ m_A a^2 + m_B b^2 = J_s = m i_s^2. \end{cases}$$

$$m_A = \frac{J_s}{al}; \quad m_B = \frac{J_s}{bl}; \quad m_s = m - \frac{J_s}{ab} = m \left(1 - \frac{i_s^2}{ab} \right).$$

$$m_{AS} = m_S \frac{b}{l}; \quad m_{BS} = m_S \frac{a}{l}.$$

$$m_A = m \frac{b}{l}; \quad m_s = m \frac{a}{l}.$$



Цел на уравнивяването и условия за уравнивесеност

$$\Phi_{\Sigma x} = 0; \Phi_{\Sigma y} = 0; \Phi_{\Sigma z} = 0; M_{\Sigma x}^{\Phi} = 0; M_{\Sigma y}^{\Phi} = 0; M_{\Sigma z}^{\Phi} = 0.$$

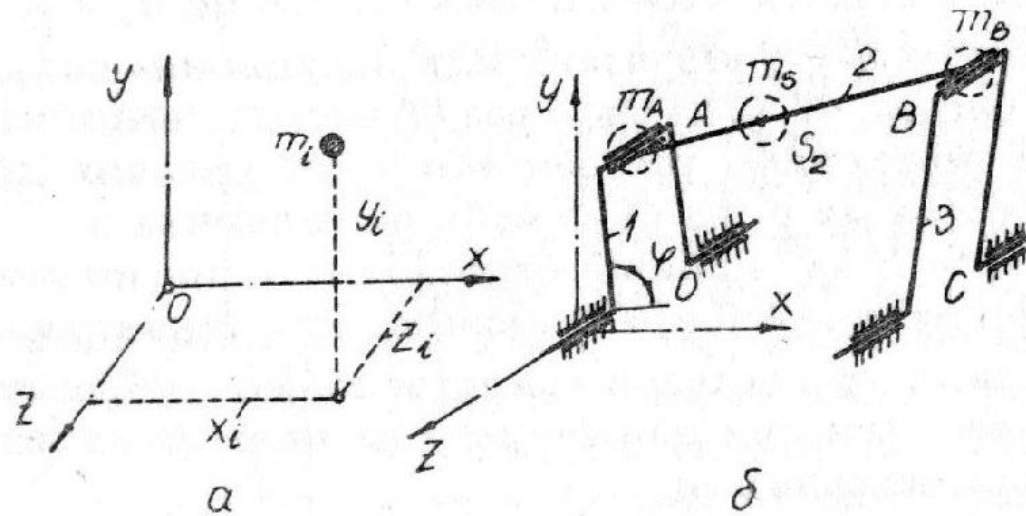
$$\Sigma \vec{F}_i = m \vec{a}_s = -\vec{\Phi}_{\Sigma}.$$

$$x_s = \text{const}; y_s = \text{const}; z_s = \text{const}.$$

$$\frac{d}{dt} K_x = M_x^{F_i}; \quad \frac{d}{dt} K_y = M_y^{F_i}; \quad \frac{d}{dt} K_z = M_z^{F_i}.$$

Векторната производна на главния момент на кол. на движение е равна на главния момент на външните сили.

$$K_x = \text{const}; \quad K_y = \text{const}; \quad K_z = \text{const}.$$



$$\left\{ \begin{aligned} K_x &= \Sigma m_i z_i V_{yi} = \Sigma m_i z_i \frac{dy_i}{dt} = \frac{dy_i}{dt} \Sigma m_i z_i y_i = \frac{d}{dt} J_{yz} \\ K_y &= \Sigma m_i z_i V_{xi} = \Sigma m_i z_i \frac{dx_i}{dt} = \frac{d}{dt} \Sigma m_i z_i x_i = \frac{d}{dt} J_{xz}. \end{aligned} \right.$$

$$K_x = \frac{d}{dt} J_{yz} = \text{const}; \quad K_y = \frac{d}{dt} J_{xz} = \text{const}.$$

$$J_{yz} = C_1 t + D_1; \quad J_{xz} = C_2 t + D_2,$$

$$J_{xz} = \text{const}; \quad J_{yz} = \text{const}$$

$$x_s = \text{const}; \quad y_s = \text{const}; \quad J_{xz} = \text{const}; \quad J_{yz} = \text{const}.$$

Уравновесяване на равнинни механизми

$$\left\{ \begin{array}{l} m_{O_1} + m_{A_1} m_1; m_{O_1} \cdot \overline{OS}_1 = m_{A_1} \cdot \overline{AS}_1; \\ m_{A_2} + m_{b_2} = m_2; m_{A_2} \cdot \overline{AS}_2 = m_{b_2} \cdot \overline{BS}_2; \\ m_{B_3} = m_3. \end{array} \right.$$

$$m_{O_1}, m_A = m_{A_1} + m_{A_2} \quad \text{и} \quad m_B = m_{B_2} + m_3,$$

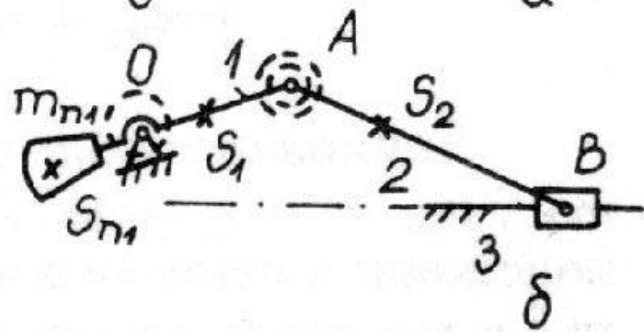
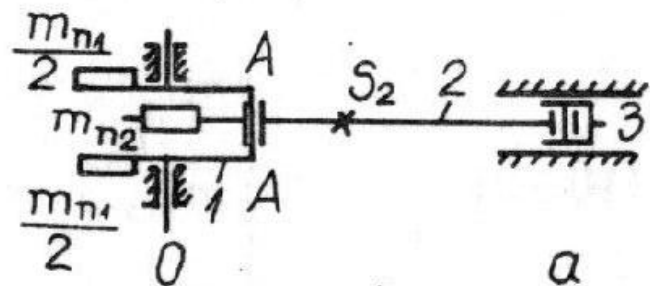
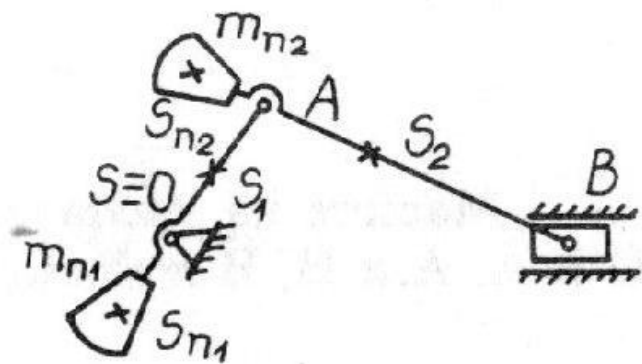
$$m_{\Pi_2} \cdot \overline{AS}_{\Pi_2} = m_B \overline{AB}$$

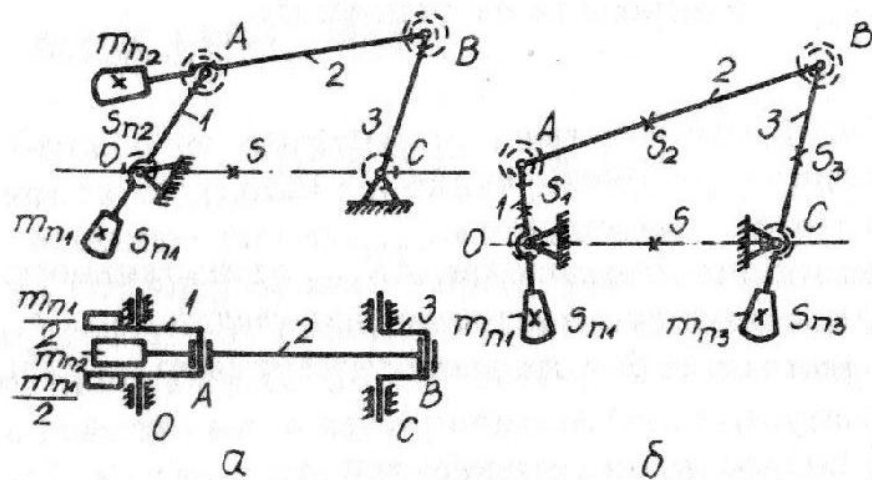
$$m_A^\Sigma = m_B + m_A + m_{\Pi_2}$$

$$m_{\Pi_1} \overline{OS}_{\Pi_1} = m_A^\Sigma \overline{OA}.$$

$$m_0^\Sigma = m_{\Pi_1} + m_{O_1} + m_A^\Sigma$$

$$= m_1 + m_2 + m_3 + m_{\Pi_1} + m_{\Pi_2}.$$





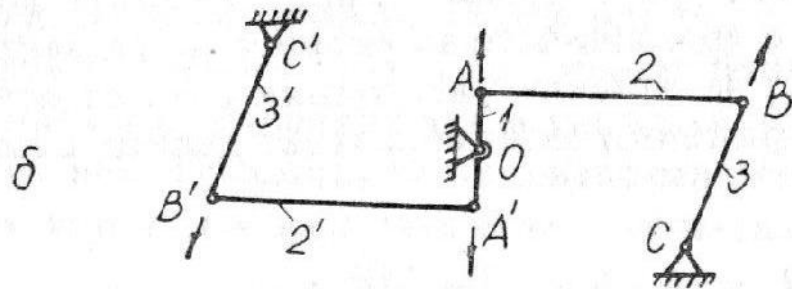
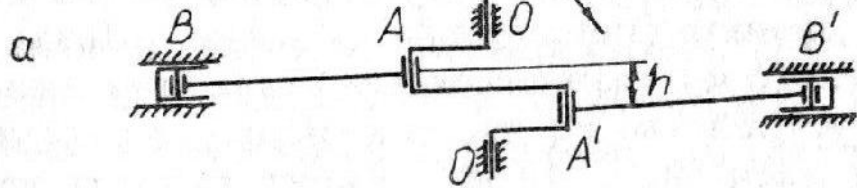
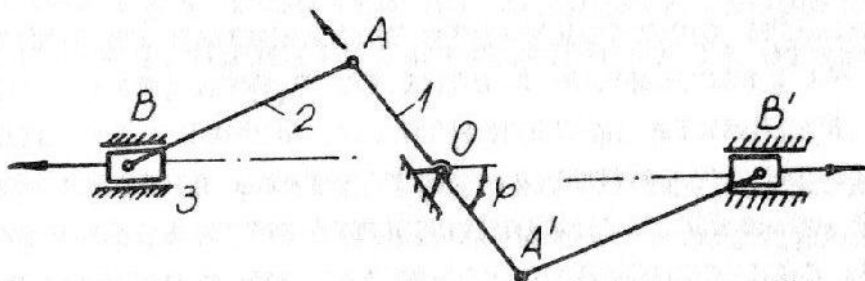
$$\begin{cases}
 m_{O_1} + m_{A_1} = m_1; & m_{O_1} \cdot \overline{OS_1} = m_{A_1} \cdot \overline{AS_1}; \\
 m_{A_2} + m_{B_2} = m_2; & m_{A_2} \cdot \overline{AS_2} = m_{B_2} \cdot \overline{BS_2}; \\
 m_{B_3} + m_{C_3} = m_3; & m_{B_3} \cdot \overline{BS_3} = m_{C_3} \cdot \overline{CS_3}.
 \end{cases}$$

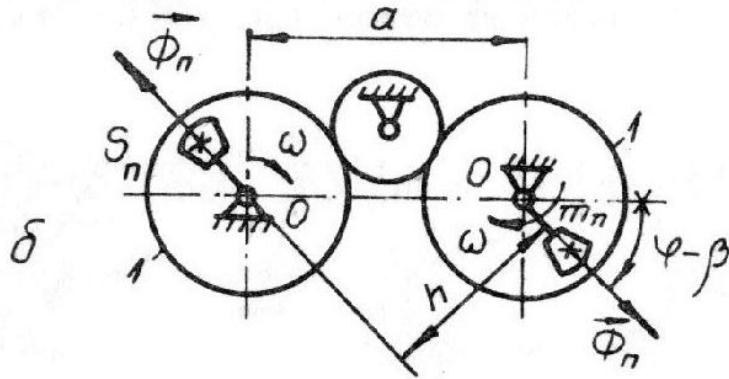
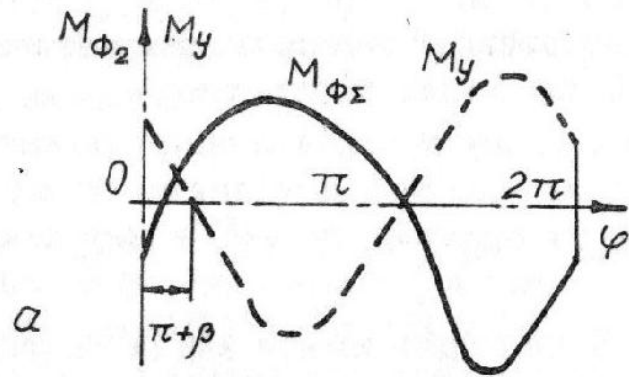
$$m_{\Pi 1} \cdot \overline{OS_{\Pi 1}} = m_A \cdot \overline{OA}; \quad m_{\Pi 3} \cdot \overline{CS_{\Pi 3}} = m_B \cdot \overline{CB}.$$

$$m_O^{\Sigma} = m_{\Pi 1} + m_{O_1} + m_A; \quad m_C^{\Sigma} = m_{\Pi 3} + m_{C_3} + m_B.$$

$$m_0^\Sigma \cdot \overline{OS} = m_C^\Sigma (\overline{OC} - \overline{OS}).$$

$$m^\Sigma = m_0^\Sigma + m_C^\Sigma = m_1 + m_2 + m_3 + m_{\Pi_1} + m_{\Pi_3}.$$





$$M_y = \Phi_{\Pi} h = \omega^2 m_{\Pi} \overline{OS}_{\Pi} a \sin(\varphi - \beta).$$

Уравновесяване на ротори

$$d\Phi^n = \rho\omega^2 dm; \quad d\Phi^t = \rho\epsilon dm.$$

$$d\Phi_x = \omega^2 \rho \cos \alpha dm - \epsilon \rho \sin \alpha dm = \omega^2 x dm - \epsilon y dm,$$

$$d\Phi_y = \omega^2 \rho \sin \alpha dm + \epsilon \rho \cos \alpha dm = \omega^2 y dm + \epsilon x dm,$$

$$d\Phi_z = 0,$$

$$dM_x = d\Phi_y z = \omega^2 y z dm + \epsilon x z dm,$$

$$dM_y = -d\Phi_x z = -\omega^2 x z dm + \epsilon y z dm,$$

$$dM_z = d\Phi_x y - d\Phi_y x = -\epsilon(x^2 + y^2) dm.$$

$$d\Phi_x = \omega^2 \rho \cos \alpha dm - \epsilon \rho \sin \alpha dm = \omega^2 x dm - \epsilon y dm,$$

$$d\Phi_y = \omega^2 \rho \sin \alpha dm + \epsilon \rho \cos \alpha dm = \omega^2 y dm + \epsilon x dm,$$

$$d\Phi_z = 0,$$

$$dM_x = d\Phi_y z = \omega^2 y z dm + \epsilon x z dm,$$

$$dM_y = -d\Phi_x z = -\omega^2 x z dm + \epsilon y z dm,$$

$$dM_z = d\Phi_x y - d\Phi_y x = -\epsilon(x^2 + y^2) dm.$$

$$\Phi = \sqrt{\Phi_x^2 + \Phi_y^2} = m \sqrt{x_s^2 + y_s^2} \sqrt{\omega^4 + \epsilon^2} = m \rho_s \sqrt{\omega^4 + \epsilon^2},$$

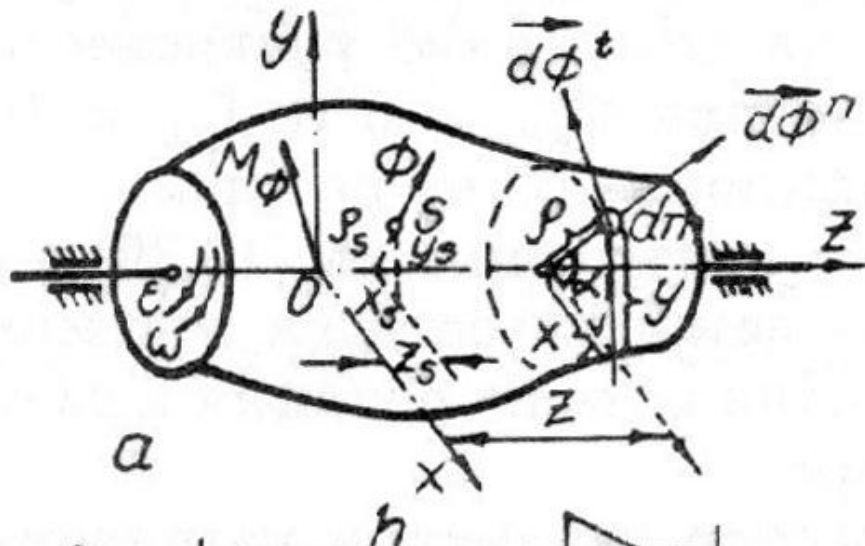
$$M_\Phi = \sqrt{M_x^2 + M_y^2} = \sqrt{J_{xz}^2 + J_{yz}^2} \sqrt{\omega^4 + \epsilon^2},$$

$$\Phi = 0 \quad \text{и} \quad M_\Phi = 0.$$

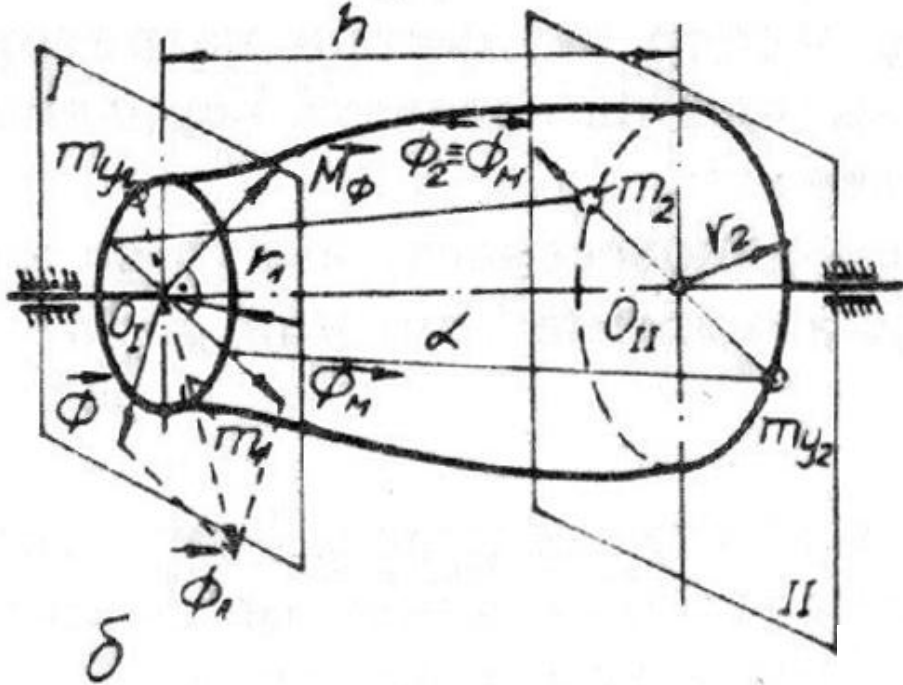
$$\Delta_c = m\rho_s = \frac{G}{g}\rho_s.$$

$$\Delta_g = J_{\rho z} = \sqrt{J_{xz}^2 + J_{yz}^2}.$$

$$\Delta_c = m\rho_s = 0 \quad \text{и} \quad \Delta_g = \sqrt{J_{xz}^2 + J_{yz}^2} = 0.$$



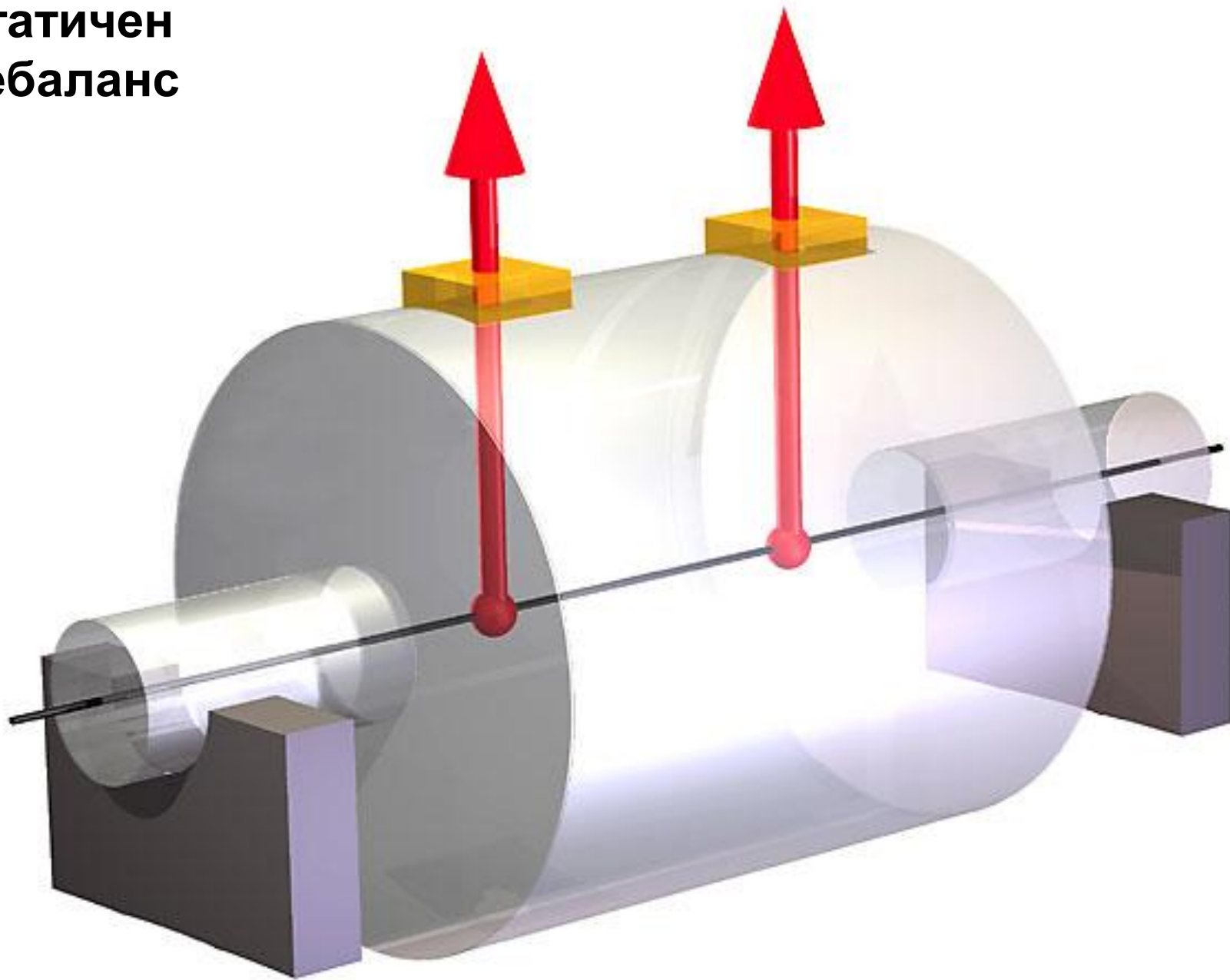
$$\Phi_M = \frac{M_\Phi}{h},$$



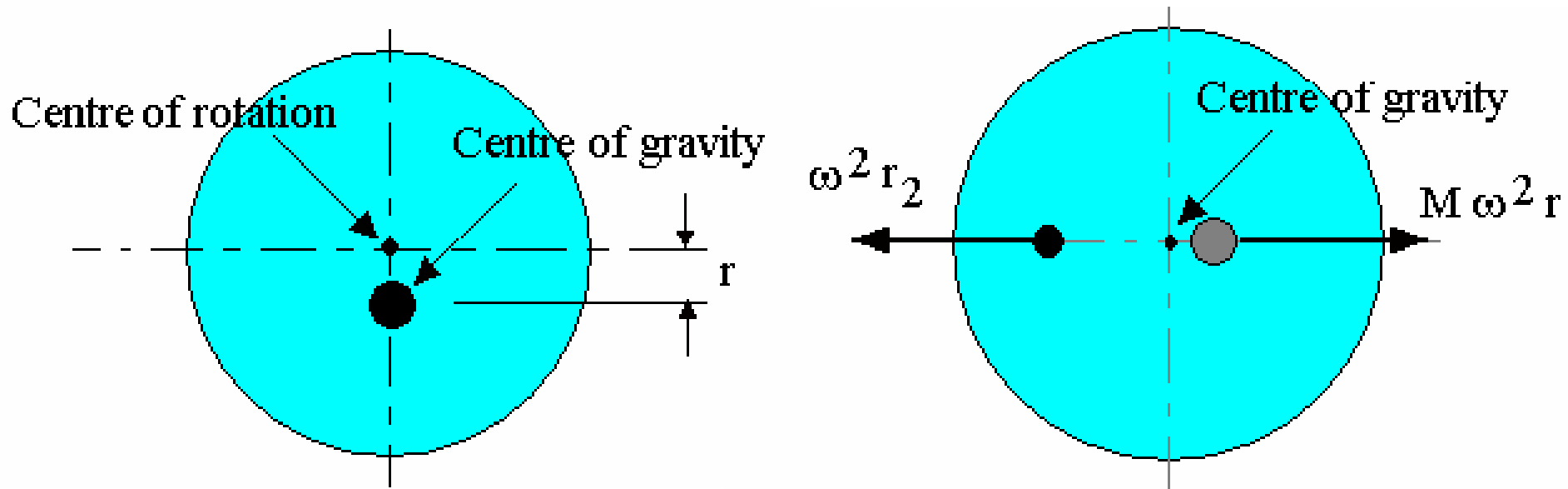
$$\vec{\Phi}_1 = \vec{\Phi} + \vec{\Phi}_M.$$

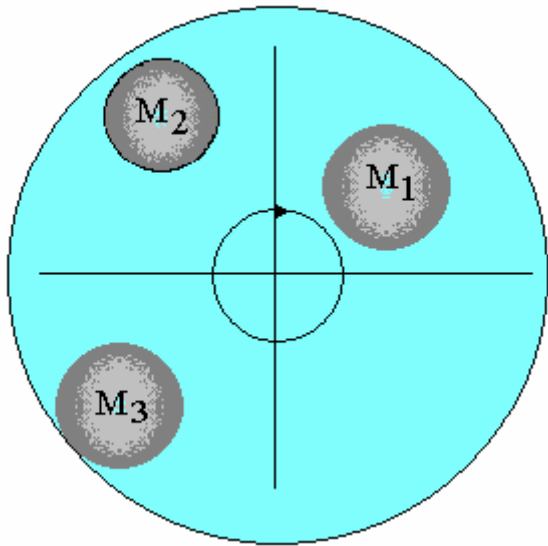
$$m_1 \equiv m_{y1} = \frac{|\vec{\Phi}_1|}{r_1 \omega^2}; \quad m_2 \equiv m_{y2} = \frac{|\vec{\Phi}_2|}{r_2 \omega^2},$$

Статичен дебаланс

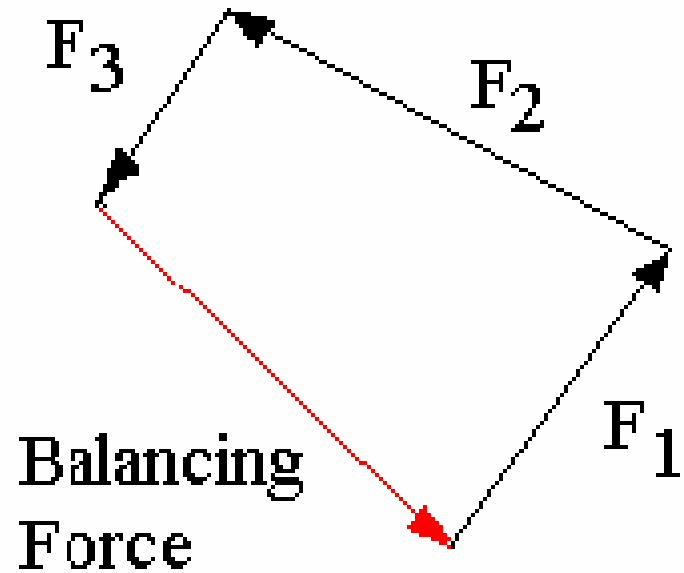
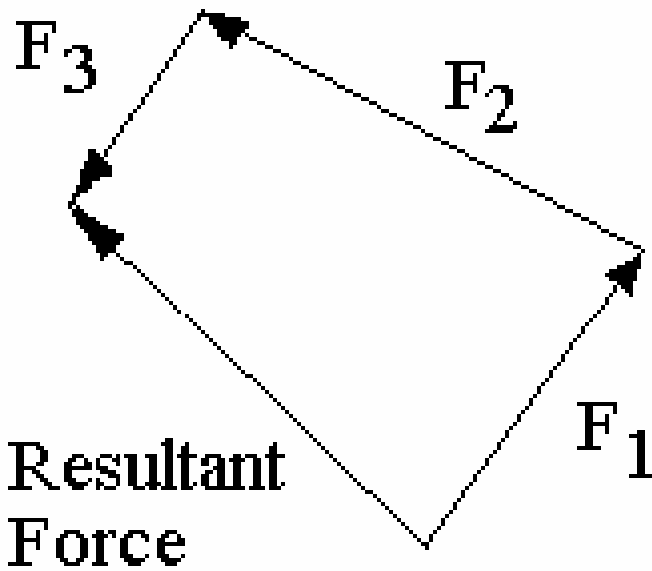
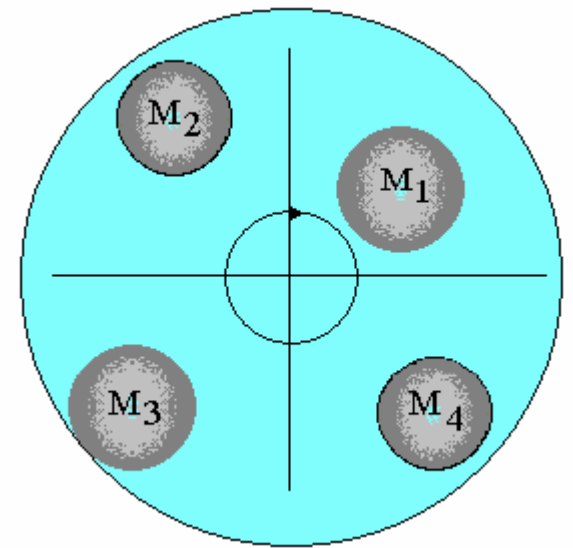


Статичен дебаланс

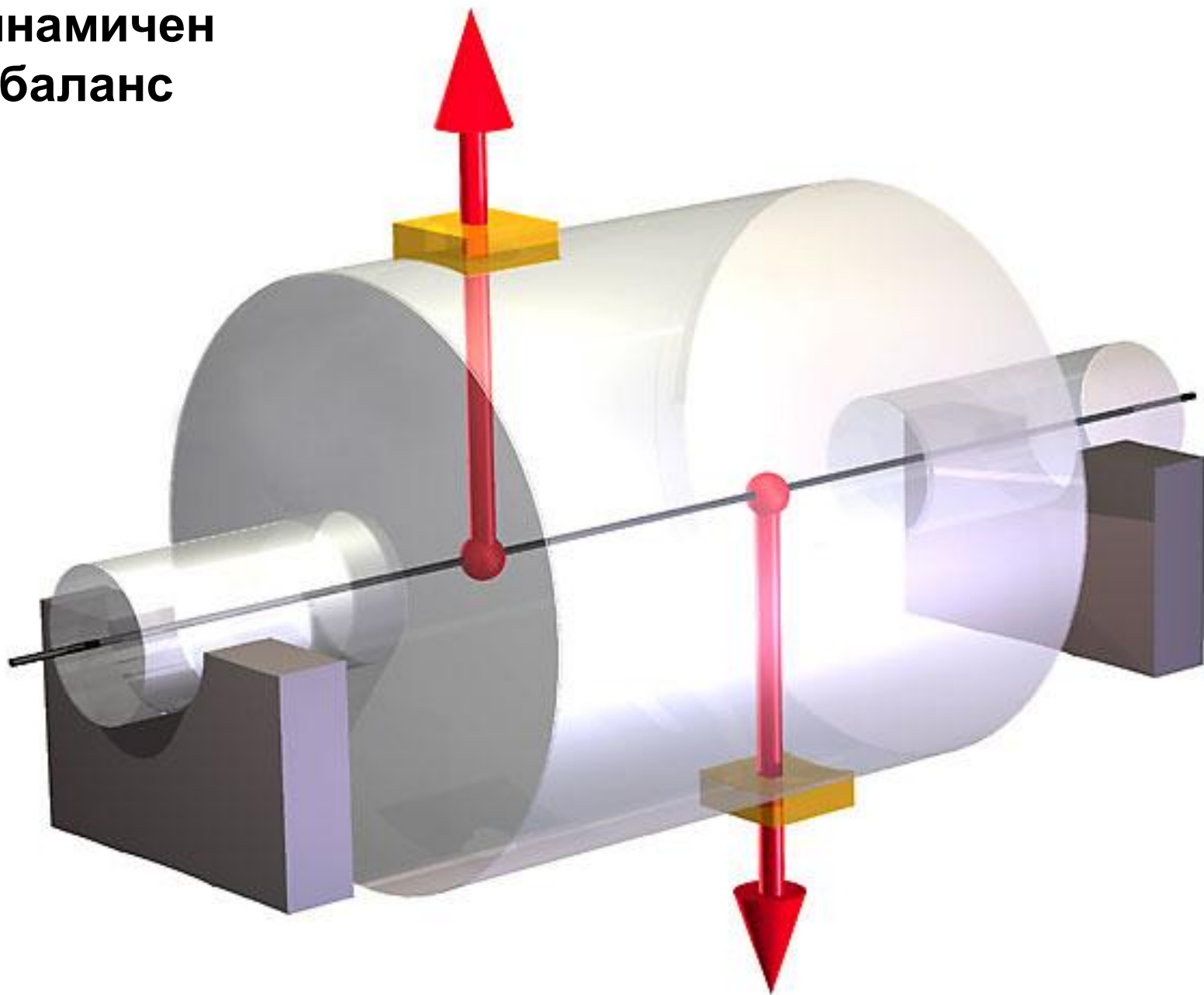




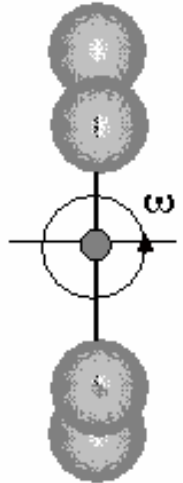
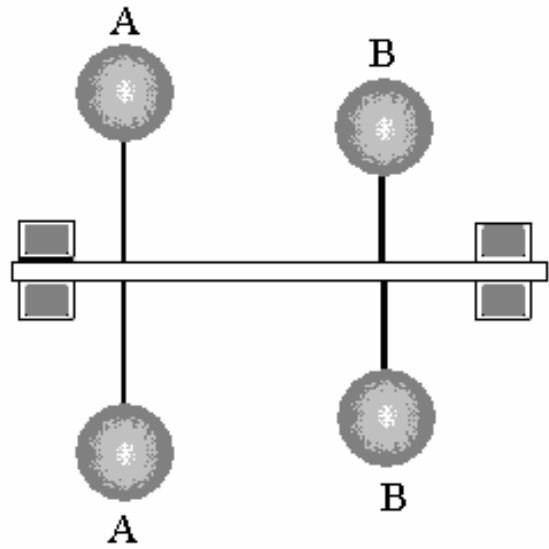
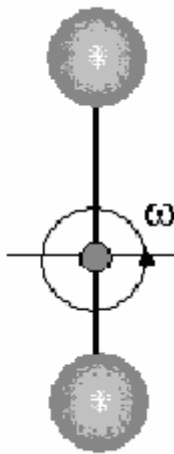
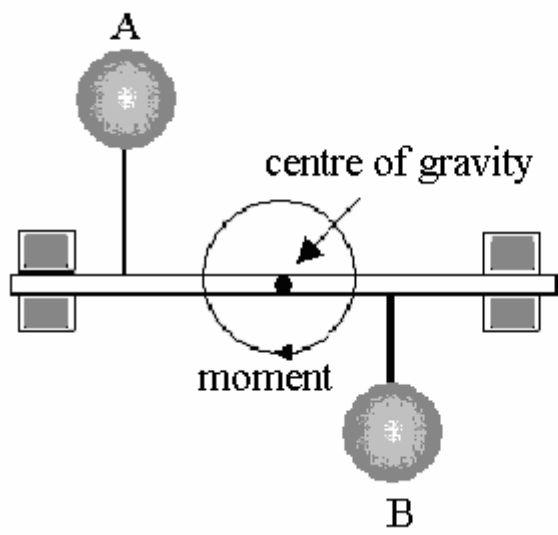
Статичен дебаланс и отстраняване



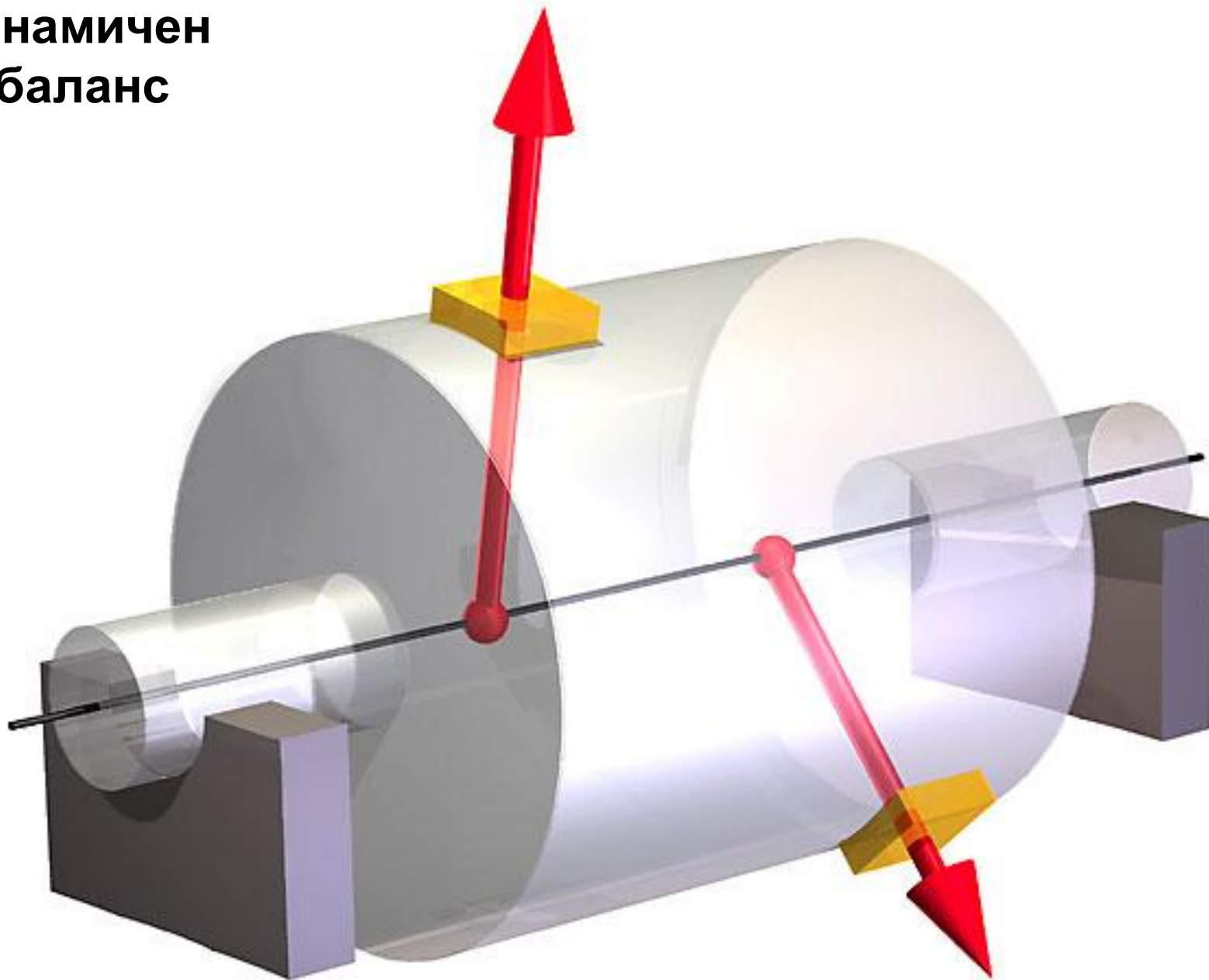
Динамичен дебаланс

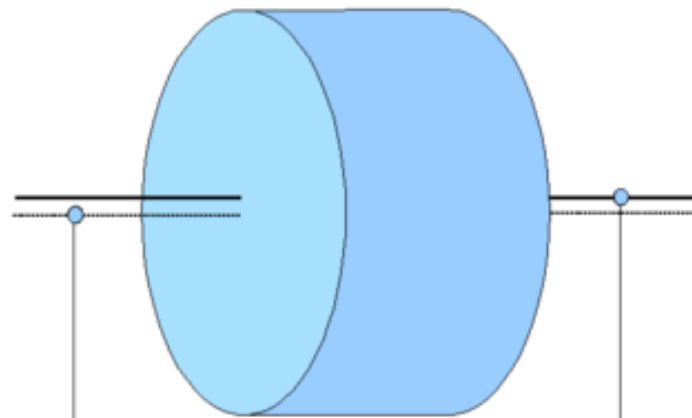


Маси в различни равнини



Динамичен дебаланс

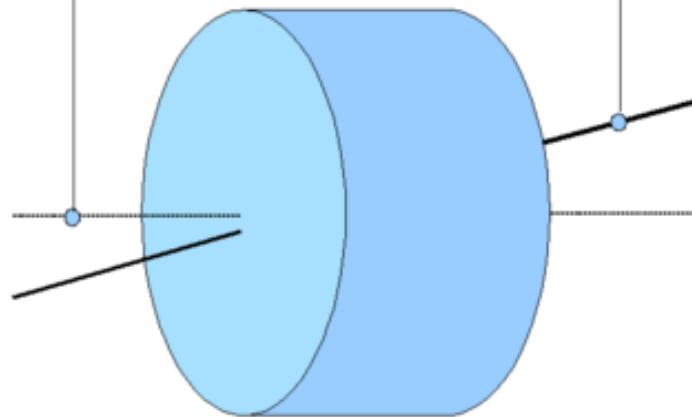




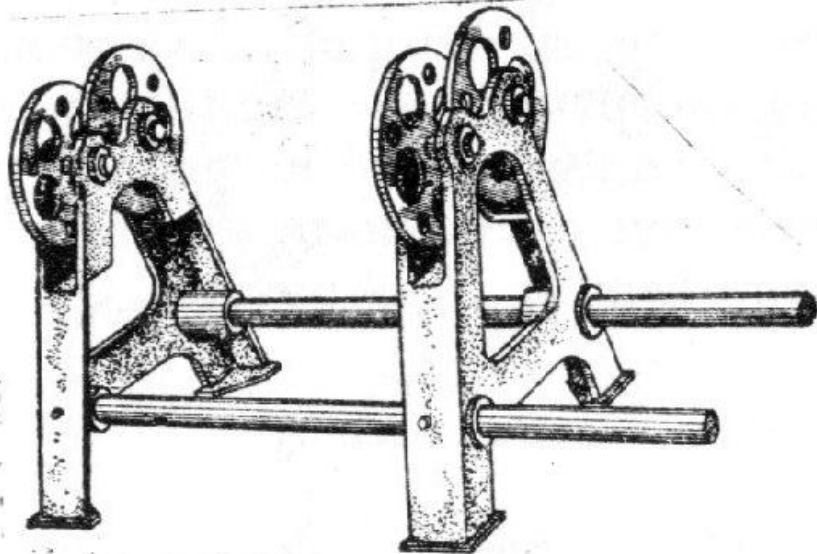
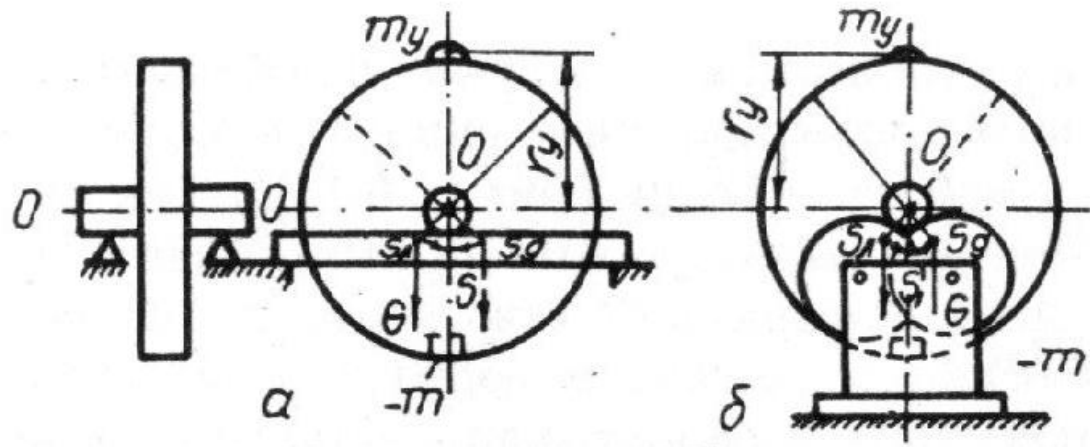
STATIC IMBALANCE

Principal Axis of Inertia

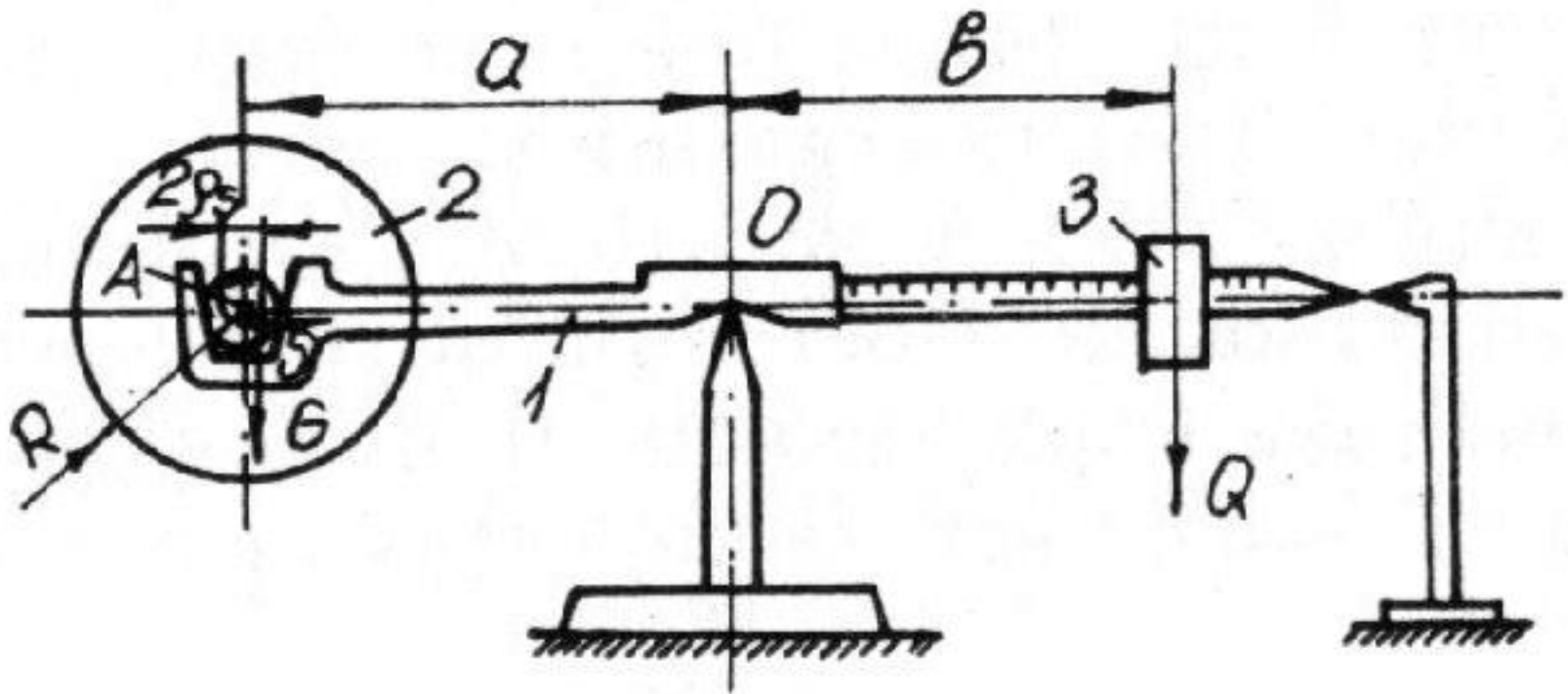
Axis of Rotation



DYNAMIC IMBALANCE



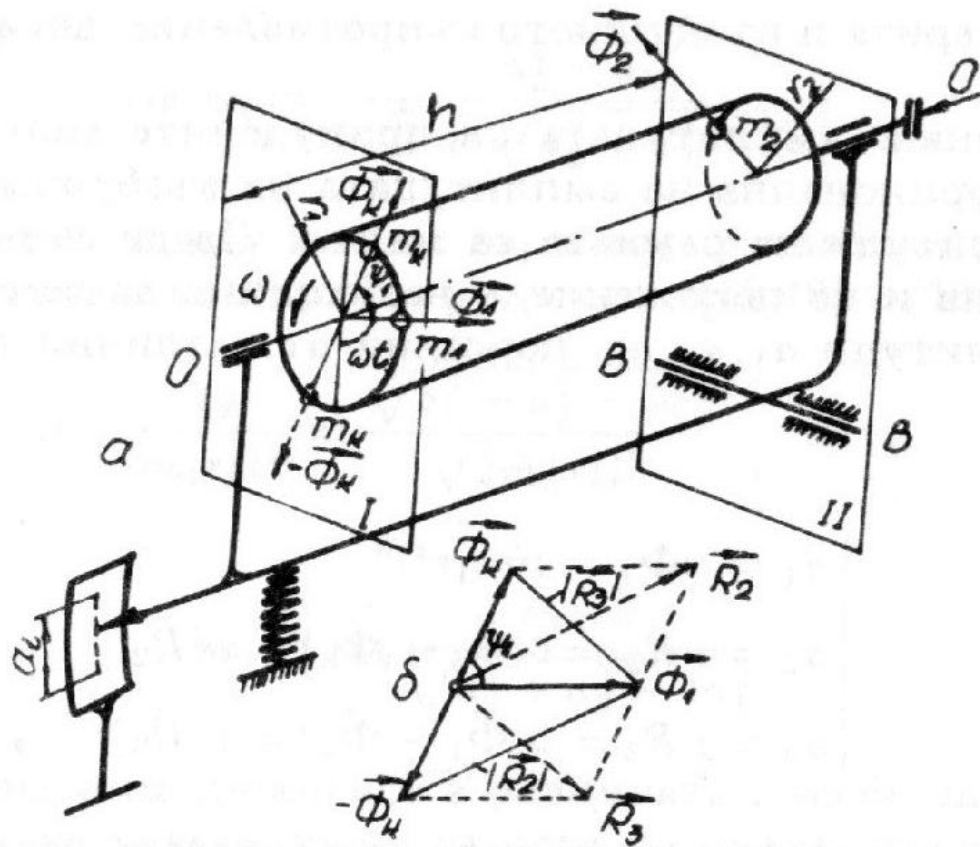
b



$$\begin{cases} M_{yp}^{\max} = Qb_{\max} = G(a + \rho_s), \\ M_{yp}^{\min} = Qb_{\min} = G(a - \rho_s). \end{cases}$$

$$\Delta_c = \frac{G}{g} \rho_s = \frac{Q}{2g} (b_{\max} - b_{\min})$$

$$m_y = -m = \frac{\Delta_c}{R}.$$



$$\begin{cases} a_1 = \nu \Phi_1 = \nu m_1 r \Omega^2 \\ a_2 = \nu R_2 = \nu |\vec{\Phi}_1 + \vec{\Phi}_k| = \nu |\vec{R}_2| \\ a_3 = \nu R_3 = \nu |\vec{\Phi}_1 - \vec{\Phi}_k| = \nu |\vec{R}_3| \end{cases}$$

$$\begin{cases} R_3^2 = \Phi_1^2 + \Phi_k^2 - 2\Phi_1\Phi_k \cos \psi_1, \\ R_2^2 = \Phi_1^2 + \Phi_k^2 - 2\Phi_1\Phi_k \cos(\pi - \psi_1). \end{cases}$$

$$\Phi_k = \frac{1}{\sqrt{2}} \sqrt{R_2^2 + R_3^2 - 2\Phi_1^2} = \frac{1}{\sqrt{2}} \sqrt{\frac{a_2^2}{\nu^2} + \frac{a_3^2}{\nu^2} - 2\frac{a_1^2}{\nu^2}};$$

$$\Phi_k = m_k r \Omega^2$$

$$\cos \psi_1 = \frac{\Phi_1^2 + \Phi_k^2 - R_3^2}{2\Phi_1 \Phi_k} = \frac{R_2^2 - R_3^2}{2\sqrt{2} \Phi_1 \sqrt{R_2^2 + R_3^2 - 2\Phi_1^2}}.$$

$$\cos \psi_1 = \frac{a_2^2 - a_3^2}{2\sqrt{2} a_1 \sqrt{a_2^2 + a_3^2 - 2a_1^2}}$$

$$\nu = \frac{a_1}{m_1 r \Omega^2} = \frac{\sqrt{a_2^2 + a_3^2 - 2a_1^2}}{\sqrt{2} m_k r \Omega^2}.$$

$$m_1 = m_k \frac{a_1 \sqrt{2}}{\sqrt{a_2^2 + a_3^2 - 2a_1^2}}.$$

$$m_{y1} r_{y1} = m_1 r.$$

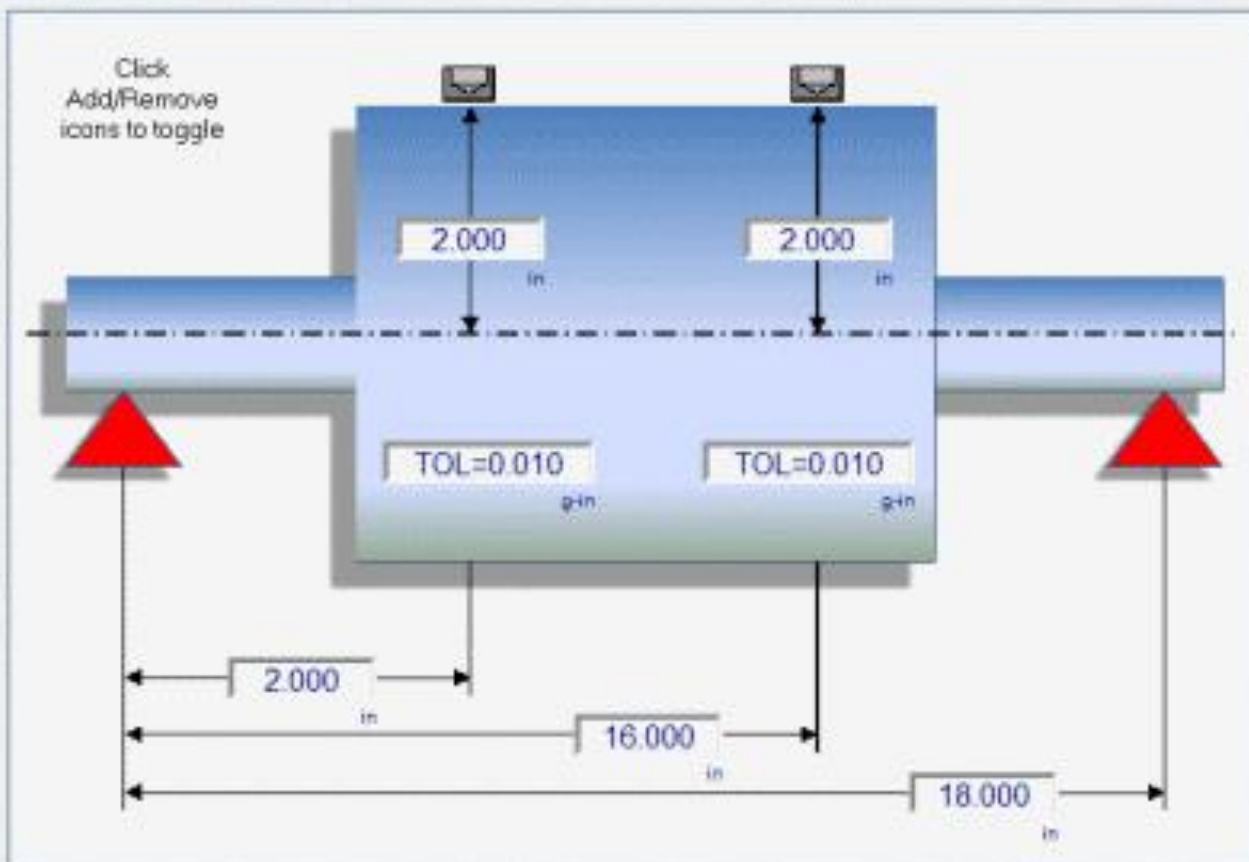


Rotor Name List TEST ROTOR

Search...
See Log
Delete Rotor

Tolerance Calculator

Journal plane
 Correction plane



Weight
gram

Distance
inch

g-in

- Radius
- Diameter
- 1-plane
- 2-plane
- Static/Couple

BalanceMaster, Inc.

Tooling Compensation: No Tooling Compensation

Key Compensation: No Key Compensation

Result Display: Polar Diagram

Start Balancing Exit